Hyperthermal Nuclear Rocket Environments Simulator



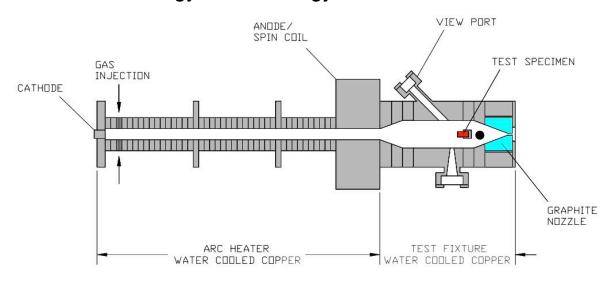
Marshall Space Flight Center

Technology Description

- 1-MW Arc-Heater Driven Hot Hydrogen Non-Nuclear Environments Simulation
 - Nuclear rocket reactor core thermal-hydraulics environments traceability
 - Long duration exposure of reactor material/fuel specimens
 - Evaluation and screening of materials and fabrication/processing techniques

Project Objectives

- Capability Demonstration Testing
- Establish Full Readiness Level
- Design Enhancements for Improved Performance & Operations
- Follow-On Technology R&D Strategy





Solid Core Nuclear Thermal Engine

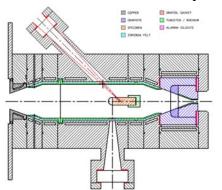
Capabilities Demonstration Testing



Technical Goals & Objectives

- Validate Hardware & Real-Time Specimen Temperature Monitoring Via Pyrometer
- Realistic Materials Specimen: W-5%Re/40%HfN CERMET
- Performance: High Temperature (2500 3000 K) & Long Duration (30 minutes)
- Operations: Multiple Tests / Fast Test Turn-Around (< 2 Hours)

Mark-I Test Fixture Assembly



Design Specifications

$$P_{\text{max}} = 500 \text{ psi}$$

$$T_{\text{max}} = 3500 \text{ K}$$

$$\dot{m} = 5$$
 to 10 g/sec

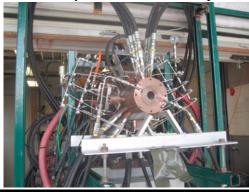
$$d_{bore} = 2.5$$
 inches

$$d_{ports} = 1.5$$
 cm

$$d_{\text{specimen}} = 0.5 \text{ inches}$$

$$l_{\text{specimen}} = 1.0$$
 inches

Full-Up Simulator Assembly



Pre-Test CERMET Specimen







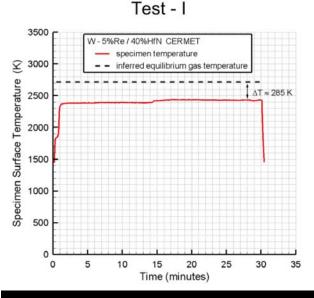
Demonstration Testing Results

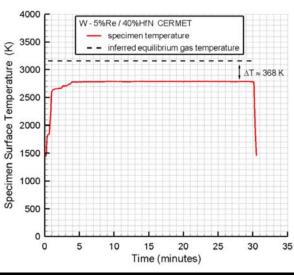


Observations & Lessons Learned

- Hardware Met or Exceeded All Design Requirements
- Excellent Set-Point Operational Stability
- Reliable & Accurate Real-Time Specimen Temperature Monitoring
- Radiation Heat Losses Higher Than Anticipated
- Unexpected Ternary Eutectic Melting of W-5%Re/40%HfN CERMET Specimen
- Enhanced Mark-II Design Recommendations
- Guideposts for Follow-On Technology R&D

Pyrometer Based Specimen Temperature Monitoring Traces





Test - II

Post-Test CERMET Specimen

